

NASA FPGA Needs and Activities



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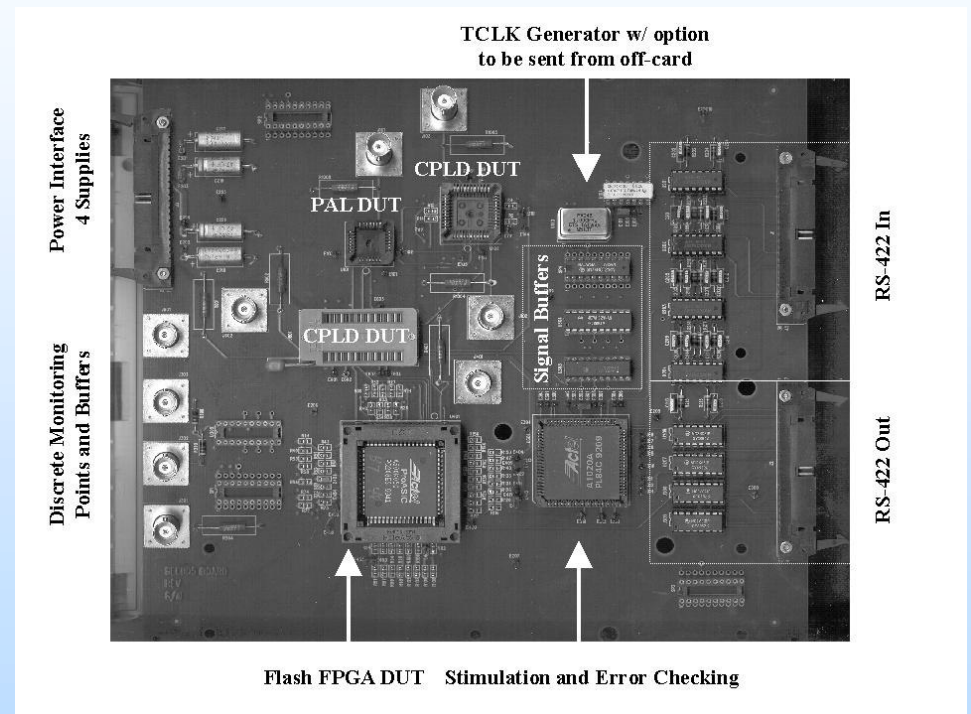
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Unclassified



Outline

- NASA Radiation Environments and Effects of Concern
- NASA Missions
 - Implications to reliability and radiation constraints
- FPGA Trade Space for NASA
- Current Usage Base
- NASA Activities in FPGAs
- FPGA Desirements
- Technical Barriers
- Summary Comments



Typical SEE Test Board for an array of programmable logic device types

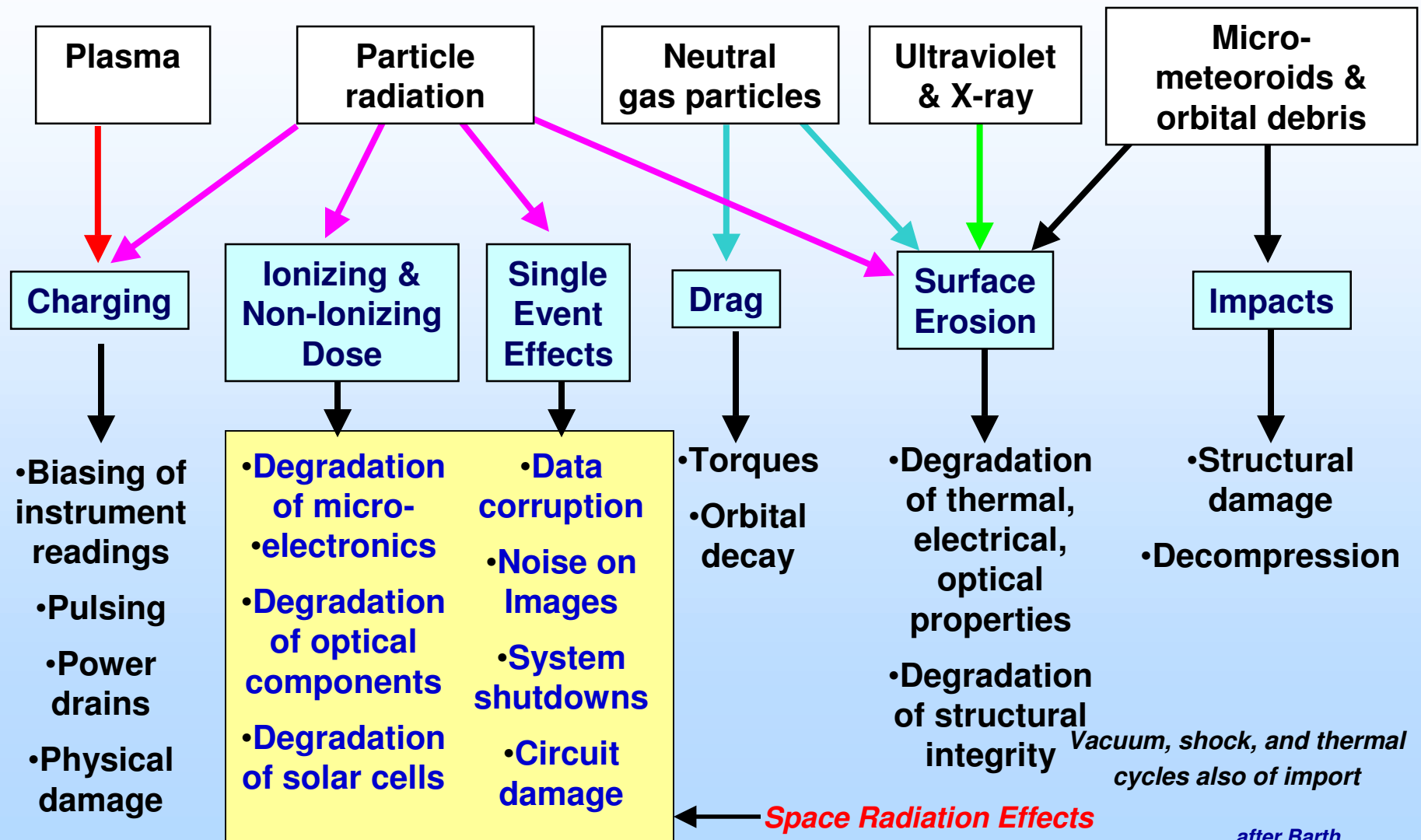
The Space Radiation Environment



***STARFISH detonation –
Nuclear attacks are not considered in this presentation***



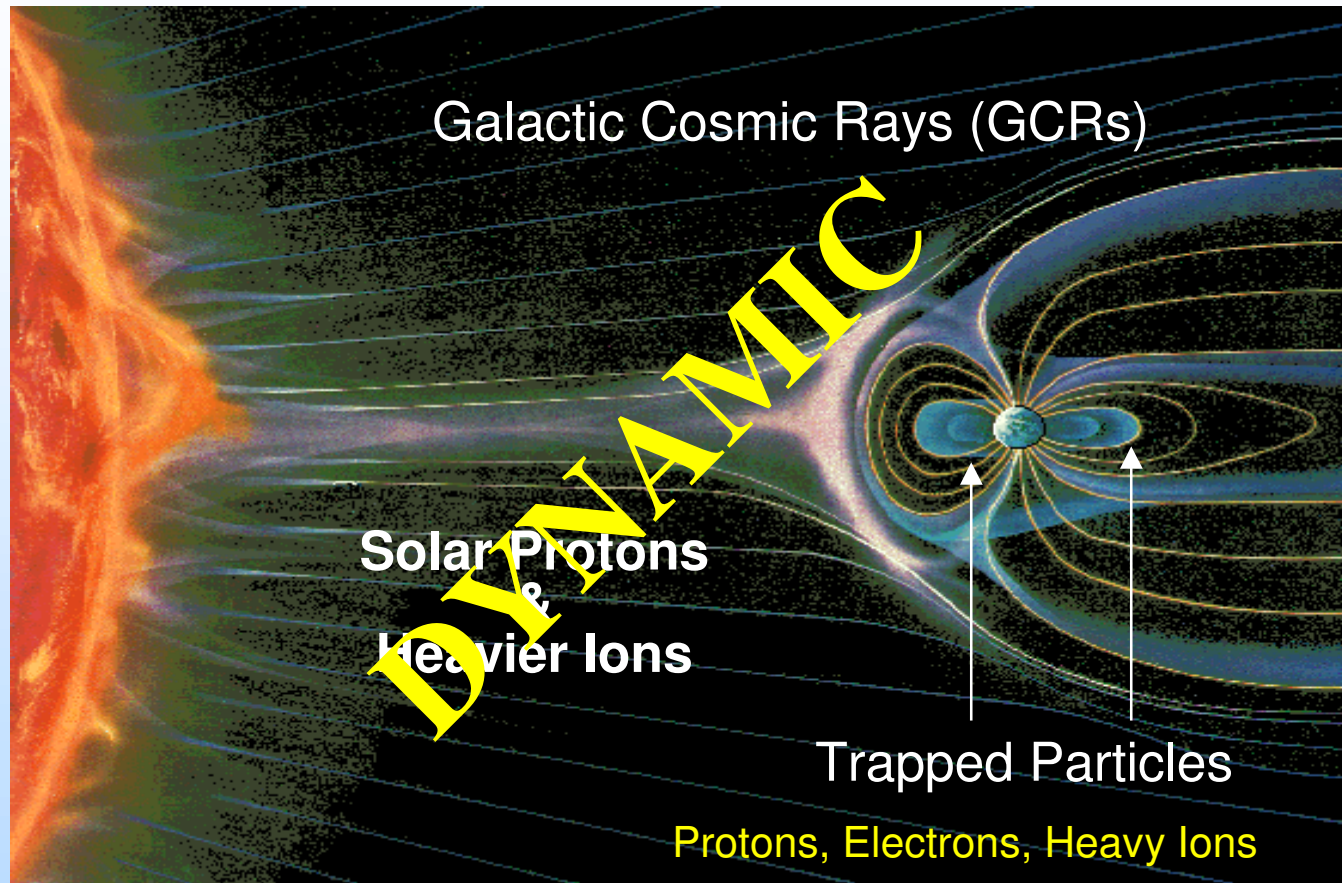
Space Environments and Related Effects





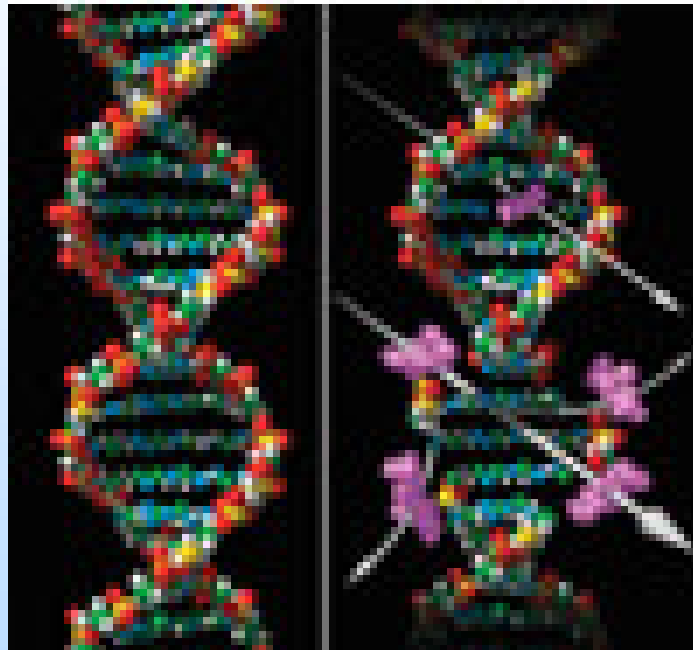
Space Radiation Environment

after
Nikkei Science, Inc.
of Japan, by K. Endo



***Deep-space missions may also see: neutrons from background
or radioisotope thermal generators (RTGs) or other nuclear source
Atmosphere and terrestrial may see GCR and secondaries***

The Effects



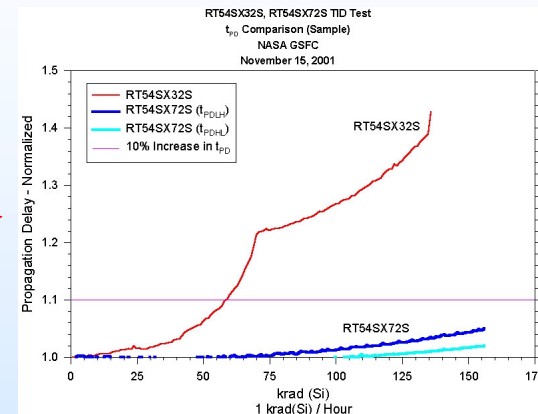
*DNA double helix
Pre and Post Irradiation
Biological effects are a key concern
for lunar and Mars missions*



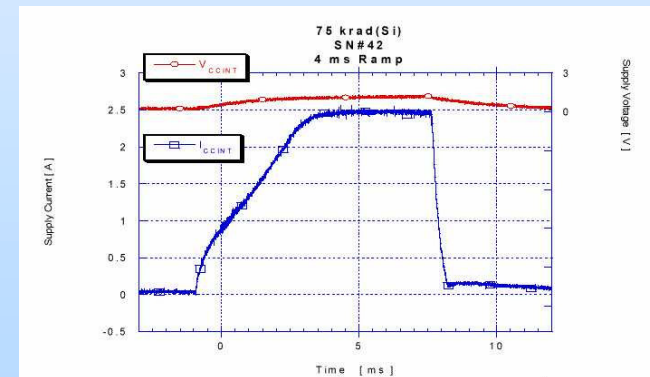
Total Ionizing Dose (TID)

- Cumulative long term *ionizing* damage due to protons & electrons
- Effects
 - Threshold Shifts
 - Leakage Current
 - Timing Changes
 - Startup Transient Current
 - Functional Failures
- Unit of interest is krad (material)
- Can *partially* mitigate with shielding
 - Low energy protons
 - Electrons

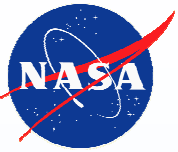
TID effects on propagation delay of a 0.25 μm FPGA.
Chart shows initial performance and that of a modified COTS rad-tolerant FPGA



Increase in startup transient current at 75 krad (Si)



TID Effects Many COTS and Modified COTS Programmable Devices



Displacement Damage (DD)

- Cumulative long term *non-ionizing* damage due to protons, electrons, and neutrons
- Effects
 - Production of defects which results in device degradation
 - May be similar to TID effects
 - Optocouplers, solar cells, CMOS, linear bipolar devices
- Unit of interest is particle fluence for each energy mapped to test energy
 - Non-ionizing energy loss (NIEL) is one means of discussing
- Shielding has some effect - depends on location of device
 - Reduce significant electron and some proton damage



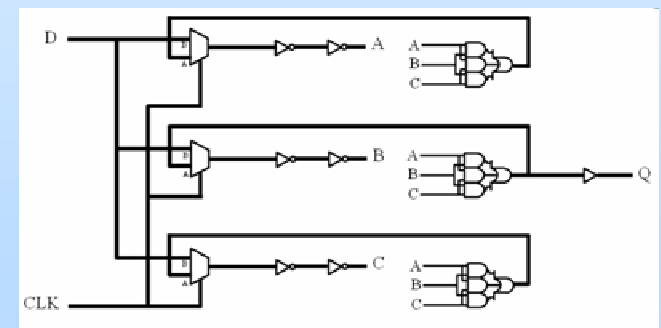
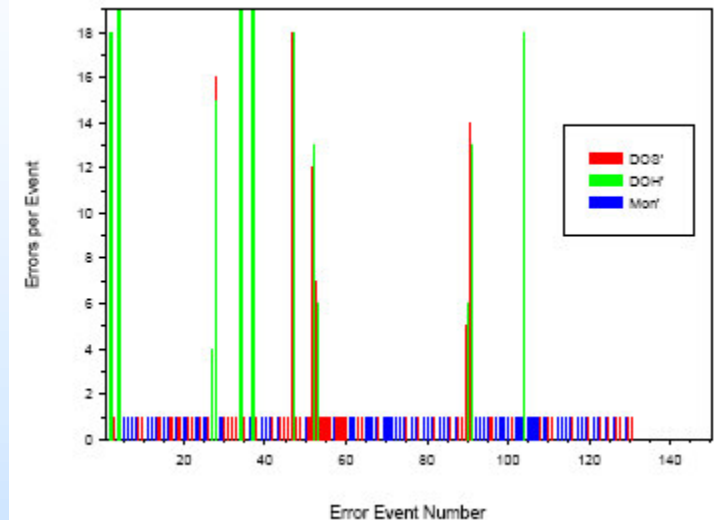


Single Event Effects (SEEs)

- An SEE is caused by a *single charged particle* as it passes through a semiconductor material
 - Heavy ions
 - Direct ionization
 - Protons for sensitive devices
 - Nuclear reactions for standard devices
 - This is similar to the soft error rate (SER) in many respects

Chart shows the number of bit errors per event in a shift error from a single SET, in this case a “clock upset.” FPGA design was subsequently modified.

- Effects on electronics
 - If the LET of the particle (or reaction) is greater than the amount of energy or *critical charge* required, an effect may be seen
 - Soft errors such as upsets (SEUs) or transients (SETs), or
 - Complete loss of control of the device, or
 - Hard (destructive) errors such as latchup (SEL), burnout (SEB), or gate rupture (SEGR)
- Severity of effect is dependent on
 - Type of effect
 - System criticality



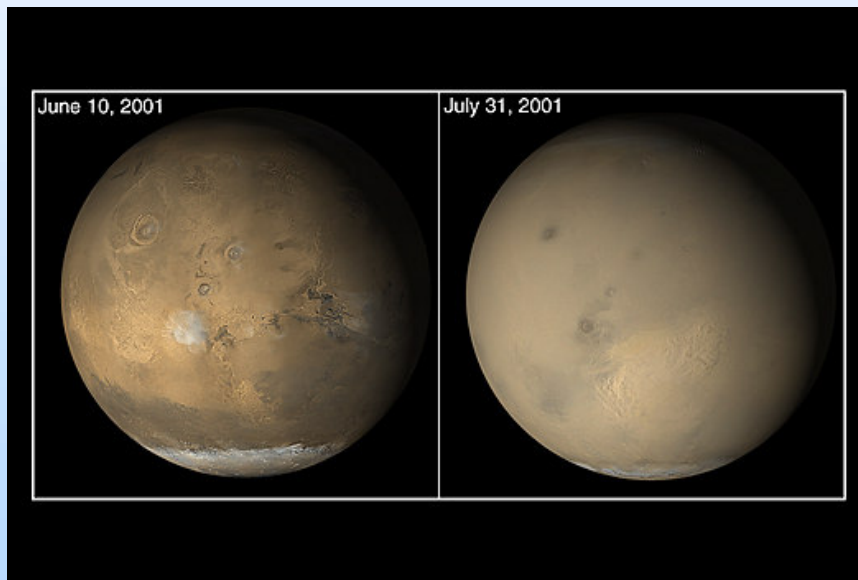
NASA designed SEU hard latch for FPGAs



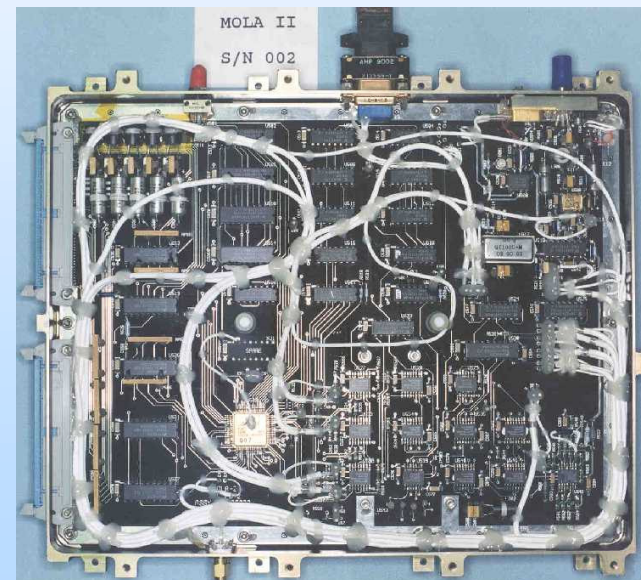
NASA Missions –

A Wide Range of Needs

- NASA typically has over 200 missions in some stage of development
 - Range from balloon and short-duration low-earth investigations to long-life deep space
 - Robotic to Human Presence
- Radiation and reliability needs vary commensurately
 - $5 \text{ krad (Si)} \leq \text{TID} \leq 100 \text{ krad (Si)}$, for $\geq 95\%$ of all mission



Mars Global Surveyor
Dust Storms in 2001



Use of FPGA in laser altimeter electronics
when upgraded from Mars Explorer
implementation.



Implications of NASA Mix to Radiation Requirements

- Prior to the new Vision for Space Exploration (re: Moon and Mars)
 - >95% of NASA missions required 100 krad (Si) or **less** for device total ionizing dose (TID) tolerance
 - Single Event Effects (SEEs) is a prime driver
 - Sensor hardness also a limiting factor
 - Many missions could accept risk of anomalies as long as recoverable over time
- Implications of the new vision are still TBD for radiation and reliability specifics, however,
 - Long-duration missions such as permanent stations on the moon require long-life high-reliability for infrastructure
 - Reliability will be the driver for FPGAs
 - Diverse technologies for manned Mars missions
 - Human presence requires conservative approaches to reliability and begets Radiation Protection Strategies
 - *Drives stricter radiation tolerance requirements and fault tolerant architectures*
 - Nuclear power/propulsion changes radiation issues (TID and displacement damage)



Lunar footprint
*Courtesy of
NASA archives
(Note some Apollo
hardware is still
functioning on the Moon
and used by scientists)*



NASA Trade Space for FPGA Usage

- **NASA tends to build one-of-a-kind instruments (or at best, a few copies) and spacecraft**
 - **ASICs are used primarily in NASA when:**
 - Performance driven: speed/size of circuit drive the need
 - Mixed-signal functions required
 - Repetitive usage of ASIC: applications that can utilize thousands of copies of the same circuit
 - We've seen up to 15,000 of the same ASIC used in a science instrument!
 - Other ASICs used in those role offer “generic” functions such as SpaceLAN, SpaceWire, Space Ethernet
 - **FPGAs are used by NASA for**
 - Standard logic replacement
 - Embedding microprocessors, memory, controllers, communications devices, in a path towards “systems on a chip.”
 - **Estimate of NASA ASIC vs. FPGA Usage**
 - NASA uses >> 10 FPGA designs for every ASIC design
 - PLD Survey currently underway



NASA Applications for FPGAs

- In essence, electronic designs may be classed into two categories for NASA space, each has critical and non-critical sections:
 - Control/Spacecraft
 - Science/Instrument
- Control applications are the heartbeat of the space system
 - Reliability, minimal downtime (re: science data loss), failure-free, etc are the drivers
 - Control applications include explosive devices
 - Without an operating spacecraft, the best instrument is useless
- Science applications are the more performance driven
 - When you are measuring the universe, you need lots of resolution, bandwidth, and memory throughput
 - We can lose some data, but we can't lose a mission
 - Tolerance is getting stricter with longer staring times, etc for instruments
- Each will be discussed with type of FPGA NASA considers



Control Applications and FPGA NASA Needs

- **Application: Control (ex., attitude control)**
- **General Needs:**
 - High-reliability
 - Radiation hardness (system must be bullet-proof)
 - Fail-safe
- **Device characteristics**
 - Small to medium size
 - 10^4 to 10^5 gates
 - Operating speed ranges from low to high (< 200 MHz system clock)
 - One-time programmable (OTP) or reprogrammable
 - Reprogrammable is often preferred for schedule and flexibility, but can complicate system design (SEU tolerance/mitigation, etc...)
 - Low to moderate power
- **Some NASA systems are using of reprogrammable devices for control**
 - Extreme care needed to prevent inadvertent deployments or other critical events



Science Applications and FPGA NASA Needs

- **Application: Science (ex., image data throughput)**
- **General Needs:**
 - Medium to High performance
 - Radiation tolerance (acceptable data losses)
 - Fail-safe
- **Device characteristics**
 - 10^4 to 10^6 gates
 - Operating speeds range from low to high (some > 200 MHz)
 - Reprogrammable
 - Preferred for flexibility to adapt algorithms for on-board processing of science data
 - Low-power desired
 - Conflicts with larger device sizes
- **Most NASA systems are still using OTP devices for radiation tolerance reasons**



Current NASA FPGA Usage

- **Primary NASA usage and plans:**
 - Actel for OTP
 - Xilinx for SEU-based reprogrammable
 - Aeroflex coming into market
 - Licensed designs and OTP technology from Quicklogic
 - Honeywell Rad-Hard Reconfigurable FPGA (RHrFPGA)
 - Sold as board-level product
 - NASA a prime funding source
 - NASA supported radiation evaluation
- **Other examples:**
 - Altera (Space Shuttle, ISS) in communications applications
 - Lucent used in GPS receivers/processors
 - PLD's used in X-vehicles (planes)



NASA R&D Activities on FPGAs

- **Reliability and Radiation Evaluation**
 - Actel 54RTSX-S Programmed Antifuse Investigation
 - Rich Katz, NASA Office of Logic Design (OLD)
 - See <http://klabs.org> for details
 - Radiation evaluation board in design for Aeroflex FPGAs
 - Xilinx Virtex-II Pro
 - Funded by Missile Defense Agency
 - Consortia with AFRL, NAVSEA
 - SEE test scheduled for Aug 16th
 - JPL represents NASA on Xilinx SEE Consortia
 - See <http://klabs.org> for other recent radiation efforts
- **Architecture**
 - Multiple efforts looking at COTS reprogrammable FPGAs
 - System architectures funded under former NASA Code R MSMT
 - MDA funding of architecture work on Xilinx Virtex-II Pro
- **Military and Aerospace Programmable Logic Devices International Conference (MAPLD)**
 - September 8-10, 2004 in Washington, DC
 - Hosted by NASA Office of Logic Design (R. Katz)
 - <http://klabs.org/mapld04>
 - richard.b.katz@nasa.gov



NASA Desirements

- **High Reliability: ≤ 10 FITs**
- **Non-volatile**
- **Reprogrammable, Unlimited Times, High-Speed, Device Sections**
- **Rad-tolerant (configuration should be radiation-hard)**
 - ≥ 100 krad (Si)
 - ≥ 75 MeV-cm²/mg SEL
 - ≥ 75 MeV-cm²/mg Damage
 - ≥ 40 MeV-cm²/mg Configuration Memory and Control Registers SEU_{TH}
 - ≥ 40 MeV-cm²/mg SEU – Control Applications
 - ≥ 15 MeV-cm²/mg SEU – Science Data Processing Applications
- **10^5 to 10^6 Gates**
- **High-speed of Operation**
- **On-chip, dual-port, block memories**
- **Multiple on-chip processors with facilities for checkpointing, restarting, comparing, and sparing**
- **I/O Modules tolerant of different voltages and standards. This is critical as other devices on-card will be of varying technologies (commercial applications tend to not have this problem to a large extent).**
- **Support for high-speed arithmetic (e.g., fast carry chains, multipliers, etc.)**
- **Simple architecture: Complexity breeds design errors and makes validation efforts “challenging.”**
- **Reliable and Accurate Software Tools – e.g. Static Timing Analyzers**
 - Guarantee both minimum and maximum bounds
 - Min/Max Clock Skew Analysis
 - Account for radiation and life effects
- **Commercially compatible architecture**
 - Use standard tool chain
 - Available “intellectual property”
 - Large designer experience base



Technical Issues

- **Reliability: MEC SX-A and SX-S Programmed Antifuse**
 - Currently undergoing intense study, evaluation, and modifications (NASA OLD/NESC, Aerospace Corp./DoD)
 - SX-SU (UMC) alternative is also being evaluated in parallel
 - SX-A used in many military weapons
- **Radiation**
 - Commercial FLASH is horrible
 - Commercial CMOS is VERY soft to SEU and may have destructive issues
 - Scrubbing, reconfiguration are okay, but not proven, and do not cover all of memory
- **Signal Integrity**
 - Programmable drive strength, slew, and impedance
 - Improved IBIS models
- **Packaging**
 - >1000 pin packages with no simple space qualification path
 - Interconnects
 - Ground bounce and V_{DD} sag
 - Additional power and ground pins
 - Capacitors internal to the package



Comments on FPGA Radiation Needs

- **NASA recommends investments in three areas**
 - **Bulletproof** device for control application
 - Reliability
 - Radiation
 - Verifiable Designs
 - **Radiation-tolerant reprogrammable device for on-board processing and non-critical control applications**
 - Compatible with commercial design tool chain
 - Goal: No radiation mitigation required
 - Supports mitigation strategies if necessary
 - **Coordinated interagency evaluation program for COTS FPGAs**
 - Radiation - Test and mitigation
 - Reliability – Test and detailed evaluation of vendor qualification
 - Intellectual Property – Library of Government-developed IP

For additional information on NASA FPGA Efforts
<http://klabs.org>